



## ***PHM for Space Applications Panel:***

# ***Space Propulsion System Wireless Sensor Technology (WST) Needs/Challenges***

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# Wireless History

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- Wireless technology was invented in the late 1800s
- Fast forward to 1962, the first communication satellite (Telstar), was launched
- In 1979, the first automatic cellular network was developed
- Wi-Fi then became popular with the public in 1997
- Decades later, wireless sensor technology (WST) is being used for many common applications: security systems, computers, sensors
- A wide variety of space propulsion systems could benefit from WST
- This presentation is scoped to needs/challenges specifically for the following:
  - **Solids**
  - **Liquids**
  - **Air-breathing Hypersonics**

# Solids

- Description: Solid rocket motors (SRMs) generate thrust from burning solid propellants (fuel and oxidizer)
- Applications:
  - **Launch Vehicle**: Solid motors are used to boost a space vehicle from a pad
  - **Missile**: SRMs provide the thrust for strategic and tactical missiles



<http://rocket.com/files/aerojet/images/media/LaunchVehAtlasV.jpg>



<http://rocket.com/files/aerojet/images/media/TacticalArmyTACMS.jpg>

# Solids (cont.)

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- WST Needs/Challenges:
  - Parameter measurement: stress, strain, temperature, humidity, acceleration, deformation (multiple measurement types from a given sensor is a plus)
  - Unique identification capability
  - Auto-ignition avoidance (energetic material compatibility / EMI requirements)
  - Long-term reliability (on the order of 40 to 50 years)
  - Tight repeatability (minimal drift is a must)
  - Robust (especially in terms of temperature and G-shock)
  - Calibrate-able (if embedded)
  - Wireless reader (usable on a strategic missile confined within silo)
  - Light-weight (grams) and small (minimal impact to assembly/performance)

**BOTTOM LINE:** Need to define **system readiness** via **propellant/liner/insulation system health status** (pre-flight focused)

# Liquids

- Description: Liquid rockets generate thrust from burning liquefied propellants (fuel and oxidizer)
- Applications:
  - **Launch Vehicle**: Liquid rockets are used to boost a space vehicle from a pad, as well as for maneuvering it in space



[http://rocket.com/files/aerojet/images/media/spaceshuttlemain1\\_0.jpg](http://rocket.com/files/aerojet/images/media/spaceshuttlemain1_0.jpg)



cp0046 15 NSS 2015 RL10 engine

# Liquids (cont.)

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- WST Needs/Challenges:
  - Parameter measurement: speed (pump), temperature, pressure, flow, acceleration, position (multiple measurement types from a given sensor is a plus)
  - Unique identification capability
  - Tight repeatability
  - Robust (especially in terms of cryogenic-to-high-temperature thermal shock)
  - Smart / local processing with data validation and calibration capabilities
  - High-speed processing capability (on the order of milliseconds for operational control)
  - Reusability

**BOTTOM LINE:** Need to define **propulsion system capability** via **overall system health status** (pre-flight, flight, and post-flight)

# *Air-breathing Hypersonics*

- Description: Hypersonic engines generate thrust from burning fuel using external ram-air (oxidizer)
- Applications:
  - **Cruise Vehicle**: Hypersonic engines provide the thrust for cruise vehicles
  - **Missile**: Hypersonic engines provide the thrust for missiles



<http://rocket.com/files/aerojet/images/media/hypersonics1.jpg>



<http://rocket.com/files/aerojet/images/media/hypersonics2.jpg>

# ***Air-breathing Hypersonics (cont.)***

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- WST Needs/Challenges:
  - Parameter measurement: position (shock-train; actuator), speed (pump), temperature, pressure, flow, acceleration (multiple measurement types from a given sensor is a plus)
  - Unique identification capability
  - Tight repeatability
  - Robust (especially in terms of high-temperature)
  - Smart / local processing with data validation and calibration capabilities
  - High-speed processing capability (on the order of milliseconds for operational control)
  - Reusability

**BOTTOM LINE:** Need to define **propulsion system capability** via **overall system health status** (pre-flight, flight, and post-flight)

# Summary

Space Propulsion System WST Needs/Challenges							
Parameter Measurements							
- Acceleration - Temperature	- Flow - Position (actuator) - Pressure - Speed (pump)	- Deformation - Humidity - Strain - Stress	- Position (shock wave)				
<b>Solids</b>	X		X		X		X
<b>Liquids</b>	X	X			X	X	
<b>Hypersonics</b>	X	X		X	X	X	

**BOTTOM LINE: Space propulsion WST needs and challenges** have been defined, now the **PHM community** is being asked to deliver